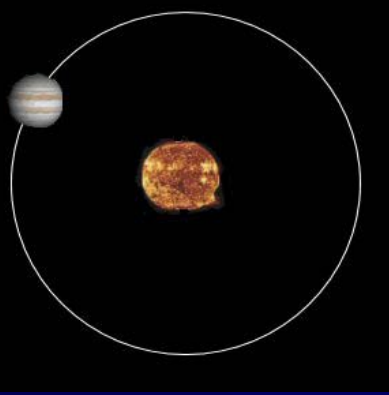
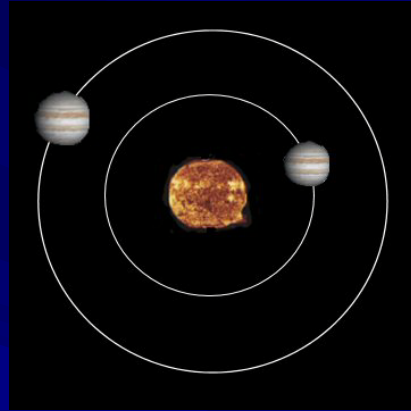


October 2012:
663 planetary systems
841 planets



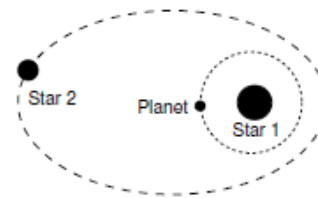
➤ **Single-Star - Single-Planet**



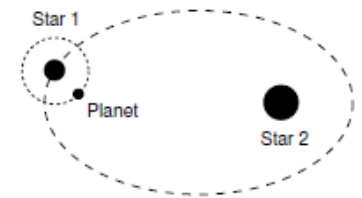
➤ **126 Multi-planet systems**
(<http://exoplanet.eu/catalog/>)

**57 planets
in binaries**

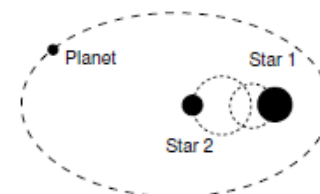
(Roell et al., 2012,
A&A)



S-Type A



S-Type B



P-Type

Two Body Problem

Interaction of two masses moving under the mutual gravitational attraction described by Newton's universal law of gravitation:

$$F = G (m_1 \cdot m_2) / d^2$$

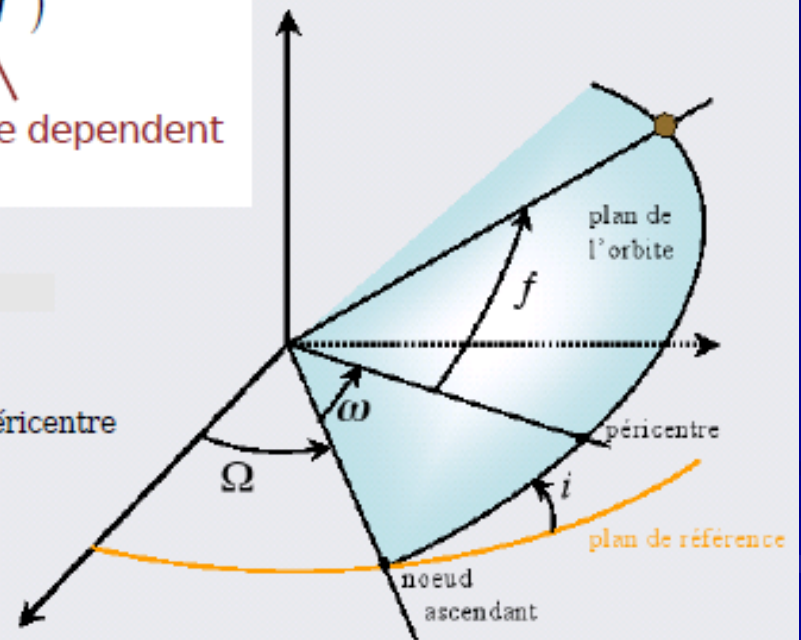
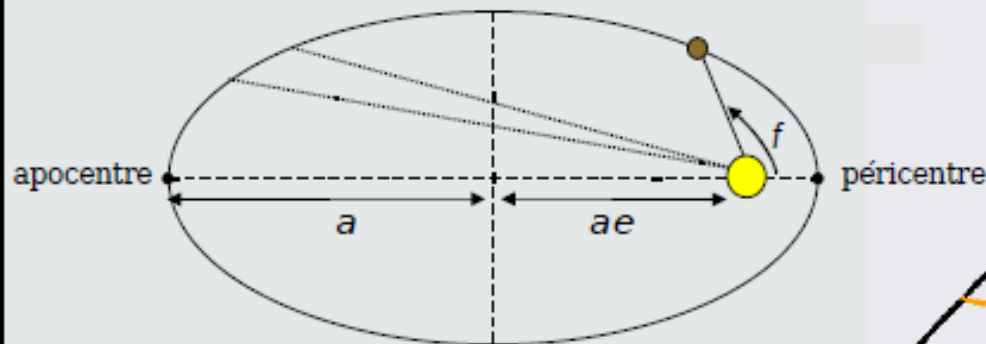
and Kepler's empirical laws of planetary motion

+ Orbital elements

The planets move in ellipses with the Sun at one focus (first Kepler's law)

$$(a, e, i, \omega, \Omega, f)$$

constants time dependent



More than 2 bodies:

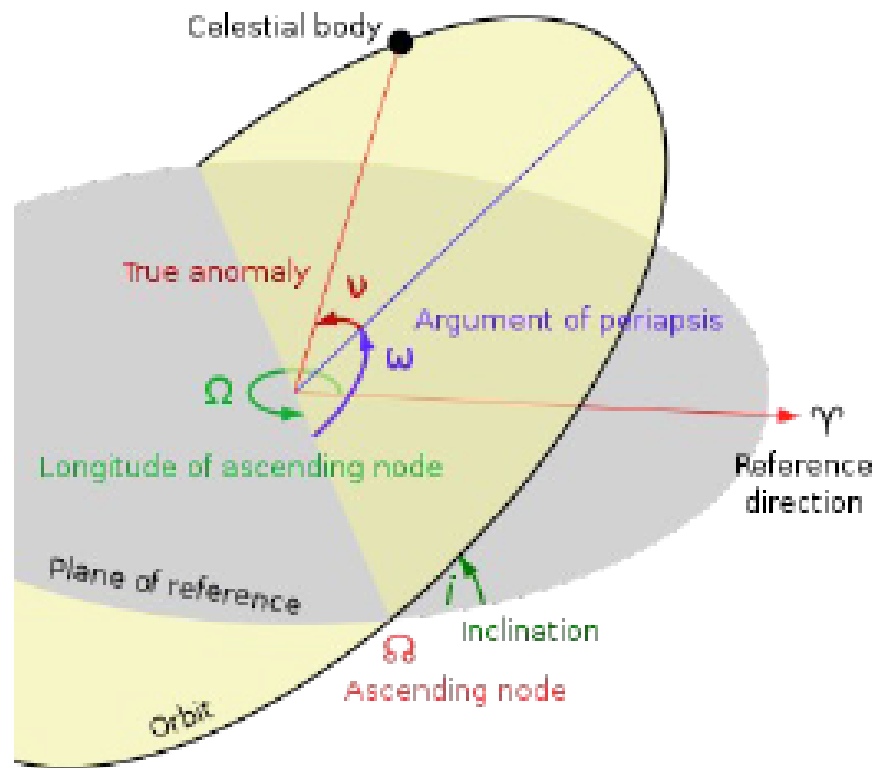
Three body problem: 18 Integrals of motion are needed
only 10 exist (Poincare)

Restricted three body problem:
2 masses + 1 massless body

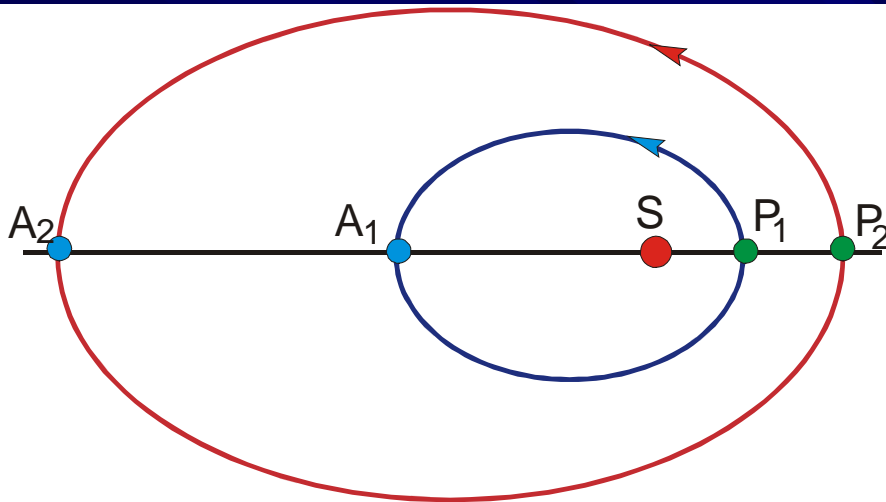
N-body problem

-> numerical Solutions

Osculating elements: $(a, e, i, \Omega, \omega, \nu)$



Resonant Motion: $n : n' = (p + q) : p$



Periastra in the same direction

Periastra in the same direction

$S - P_1 - P_2$

$S - A_1 - A_2$

$A_1 - S - P_2$

$P_1 - S - A_2$

Periastra in opposite directions

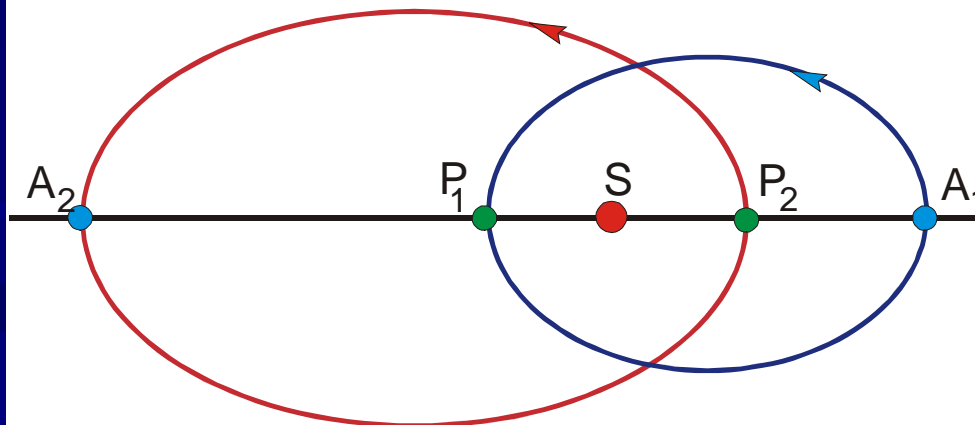
$S - P_1 - A_2$

$S - A_1 - P_2$

$P_1 - S - P_2$

$A_1 - S - A_2$

Equivalent in pairs, depending on the resonance



Periastra in opposite directions

Numerical Methods

Chaos Indicators:

Fast Lyapunov Indicator (FLI)

C. Froeschle, R. Gonczi, E. Lega
(1996)

MEGNO

RLI

Helicity Angle

LCE

Long-term numerical
integration:

Stability-Criterion:

No close encounters within
the Hill's sphere

(i) Escape time

(ii) Study of the eccentricity:
maximum eccentricity

Spacing of Planets -- Hill criterion

- Convenient rough proxy for the stability of planetary systems
- In its simplest form for planets of equal mass on circular orbits around a sun-like star

Two adjacent orbits with separation $\Delta a_i = a_{i+1} - a_i$ have to fulfill:

$$a_{i+1} = a_i + 2k R_{Hill} \quad (1)$$

$$k = 4 - 15$$

$$R_{Hill} = a_i (m / (3M))^{(1/3)}$$

→ mass-dependency

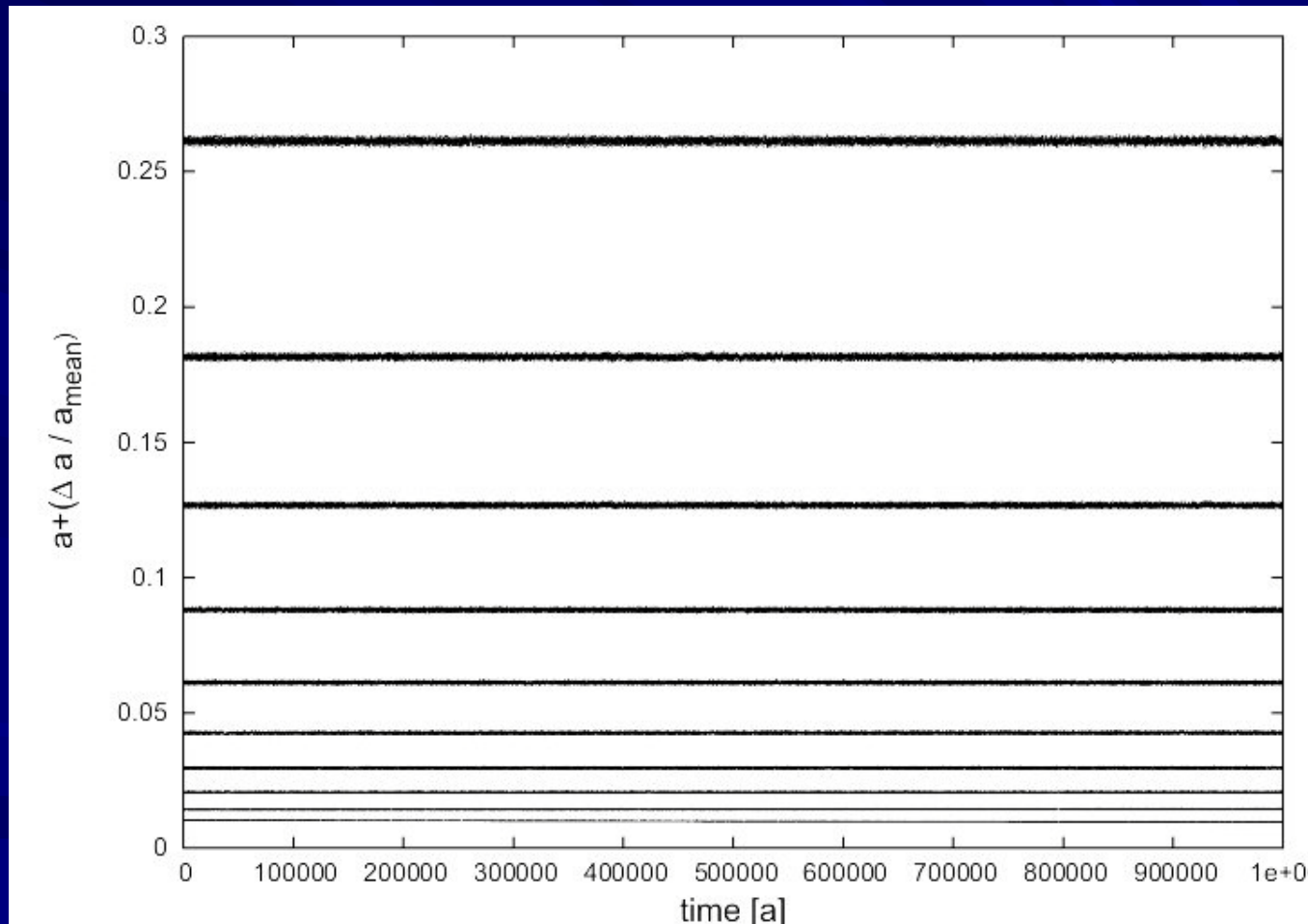
$$\frac{\Delta a_i}{a_i} \approx 2k \left(\frac{m}{3M} \right)^{1/3} \quad (2)$$

closer spacing for smaller mass planets

Numerical Study

- Fictitious compact planetary systems:
- Sun-mass star
- up to 10 massive planets (4/17/30 Earth-masses)
- $a_P = 0.01 \text{ AU} \dots\dots 0.26 \text{ AU}$
- $e, \text{incl}, \text{omega}, \text{Omega}, M: 0$

Compact planetary system using the Hill-Criterion: $m_p=1$ Earth-mass



Migration of planets

Solar System: Nice Model

NICE MODEL

Alessandro Morbidelli (OCA, France)
Kleomenios Tsiganis (Thessaloniki, Greece)
Hal Levison (SwRI, USA)
Rodney Gomes (ON-Brasil)

Tsiganis et al. (2005), *Nature* 435, p. 459

Morbidelli et al. (2005), *Nature* 435, p. 462

Gomes et al. (2005), *Nature* 435, p. 466

N-body simulations:

Sun + 4 giant planets + Disc of planetesimals

- **43 simulations** $t \sim 100$ My:

(e , $\sin I$) ~ 0.001

$a_J = 5.45$ AU , $a_S = a_J 2^{2/3} - \Delta a$, $\Delta a < 0.5$ AU

U and N initially with $a < 17$ AU ($\Delta a > 2$ AU)

Disc: $30-50 M_E$, edge at $30-35$ AU (1,000 – 5,000 bodies)

- **8 simulations** for $t \sim 1$ Gy with $a_S = 8.1-8.3$ AU

Perturbations

Mean Motion Resonances (MMRs) or Orbital Resonances

ratio of the orbital periods are integers

Secular Resonances

influence on the longitudes of perihelion or node

Initial Conditions and Computations

Planet	a (AU)	e	Inclination (deg)	ω (deg)	Ω (deg)	\mathcal{M} (deg)	Mass m (M_{\odot})
Jupiter.....	5.2028	0.0483	1.3046	275.201	100.471	183.898	0.9547907E-3
Saturn.....	9.5300	0.0533	2.4864	339.520	113.669	238.293	0.2858776E-3

Saturn: $a_{\text{sat}} = 8 \dots 11 \text{ AU}$

$m_{\text{sat}} = 1 \dots 30 m_{\text{sat}}$

Testplanets in the HZ:

$a_{\text{tp}} = 0.6 \dots 1.6 \text{ AU}$

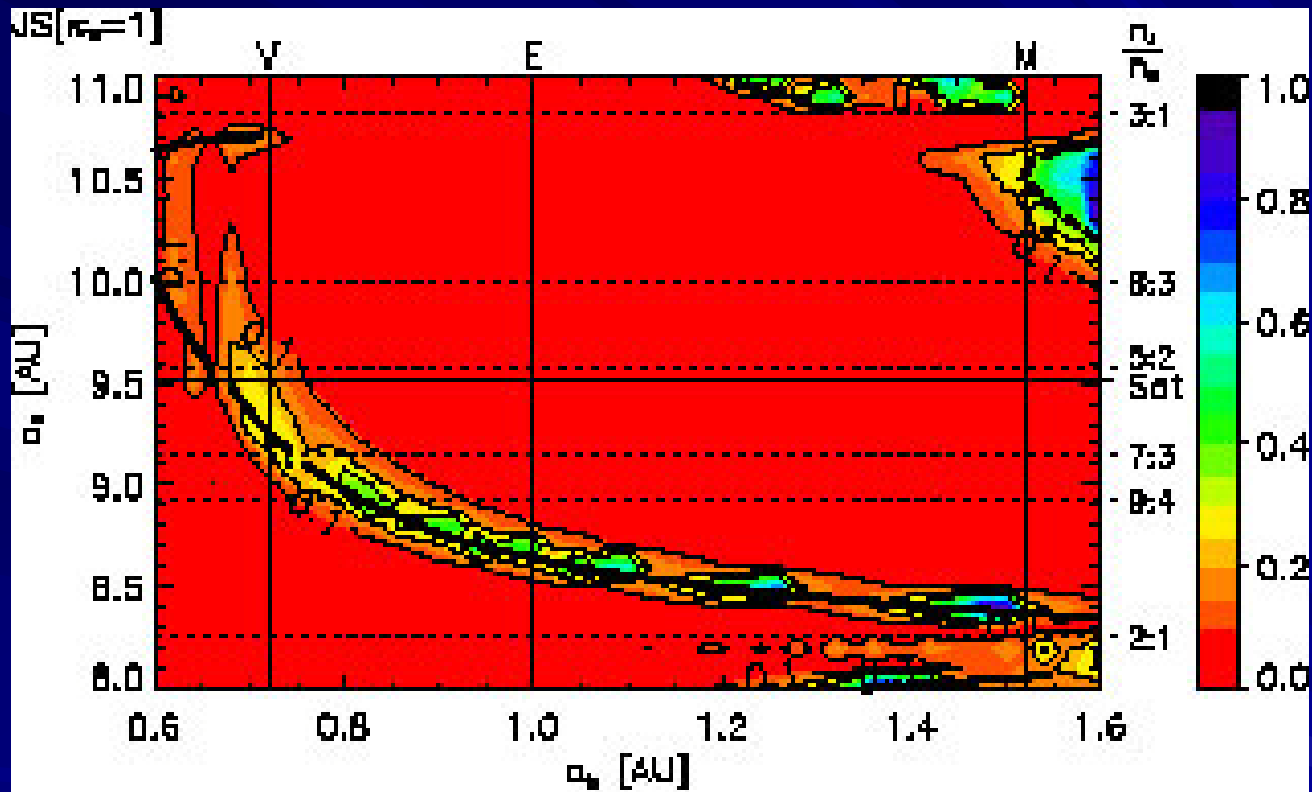
circular motion

Mercury 6 (J. Chambers)

Integration time:
20 mio years

HZ: maximum ecc.

Sun – Jupiter – Saturn



(Pilat-Lohinger et al., ApJ 2008)

Secular Perturbations:

$$g = \frac{n}{4} \left[\frac{m_J}{M_\odot} \alpha_J^2 b_{3/2}^{(1)}(\alpha_J) + \frac{m_S}{M_\odot} \alpha_S^2 b_{3/2}^{(1)}(\alpha_S) \right], \quad (1)$$

where $\alpha_J = a/a_J$, $\alpha_S = a/a_S$ (where a_J , a_S , and a are the semi-major axes of Jupiter, Saturn, and the test planet, respectively), m_J and m_S are the masses of Jupiter and Saturn, M_\odot is the mass of the Sun, and $b_{3/2}^{(1)}$ is a Laplace coefficient.

(See e.g. Murray & Dermott, Solar System Dynamics)